#### Science Mission Directorate

Technology Title: The Lightweight Integrated Solar Array and Transceiver (LISA-T)					
Affiliation:	NASA Space Technology Mission Directorate				

### Assumptions: Technology required to be at TRL 5 by 2021

# Technology Description, Current Performance Metrics, and Performance Goals

LISA-T is a launch stowed - orbit deployed small-spacecraft array with embedded lightweight power and communication devices.

- More power, less stowage volume, and less mass than state of art
- Scalable from 50W to >500W
- **Versatile** configurations pointed versus non-pointed, economic versus highest performance
- **Flexible communications** for multiple band, spherical coverage, phased arrays, or high gain design

### **Technology Development Challenges to Meet TRL Goal**

- LISA-T currently validated to **TRL6 for a LEO environment** with thermal vacuum deployment, radiation, near UV, thermal cycling, etc. testing.
- Much of the LEO testing is applicable at the component level for TRL5 in deep space.
- Zero-g parabolic flight deployments and 1 year on orbit combined environmental exposure of solar cell and antenna assemblies (MISSE10) planned to begin push towards TRL7 for LEO.
- Exploring technology demonstration opportunities to reach TRL7 (LEO); funding is challenge.

### **Contact Information**

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	rmance	Current TRL Industry State of the Art Technology Perfo						
6- LEO Array Parameter SOA Demonstrated	ed Incre	SOA I	ameter	Array Para		6- LEO	_	

6- LEO		Array Parameter	SOA	Demonstrated	Increase
5- Deep		Power generation	~80W (6U)	230.9W	288.6%
space		Stowage volume	~142kW/m <sup>3</sup>	461.8kW/m <sup>3</sup>	325.2%
		Mass	~130W/kg	378.5W/kg	291.2%
TRL By May 2021					
7- LEO 5- Deep			Planar		ni
space					

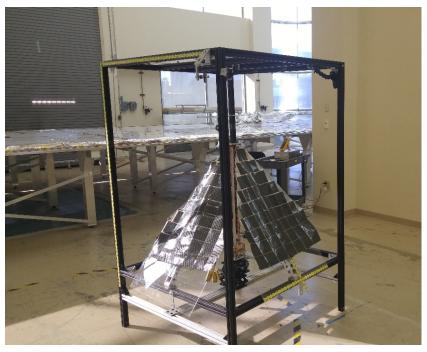
### Potential HPD Science Application (Optional)

- Highly capable small spacecraft swarms to measure space weather, radiation, etc. LISA-T can provide <u>high power with integrated comms.</u>
- >1AU small spacecraft as irradiance falls off at distance<sup>2</sup> from the sun, LISA-T can enable small spacecraft to go further from the sun.
- Reduced or completely removed solar array pointing requirements –
  Omnidirectional LISA-T can generate high power regardless of orientation to sun, enabling spacecraft to point elsewhere. E.g. do not need to slew during solar sail propulsion.

### **Additional Comments**



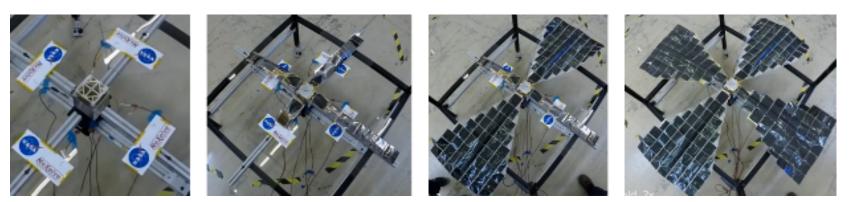
### Omnidirectional



### Planar



# Planar Deployment



### LISA-T: Solar Array Key Metrics

### Planar

Parameter	SOA (best)	IMM Array	CIGS Array
Flat Point panel			
Array power generation	~80W (6U)	230.9W	134.0W
Array stowage volume	~142kW/m <sup>3</sup>	461.8kW/m <sup>3</sup>	340.0kW/m <sup>3</sup>
Array mass	~130W/kg	378.5W/kg	250.9W/kg

Note: LISA-T power levels are scalable between ~50 and 500W.

### Omnidirectional

Parameter	SOA (best)	IMM Array	CIGS Array
Omnidirectional			
Array power generation	7.3W (3U)	101.0W	60.0W
Array stowage volume	~33W/m <sup>3</sup>	101.0kW/m <sup>3</sup>	60.0kW/m <sup>3</sup>
Array mass	~53W/kg	75.7W/kg	47.8W/kg
Generation axes	2-axis	3-axis	3-axis

## LISA-T: Antenna Key Metrics

Parameter	Endurosat Type -1	Ant Dev Microstrip	Ant Dev Helix	Clyde Space CPUT	LISA-T Edge Feed Patch	LISA-T Direct Feed Patch	LISA-T Helix
		Patch					
Antenna Gain	8.3 dBi	6 dBi	9dBi	8 dBi	6.7 dBi	7 dBi	10 dBi
Stowage Thickness	5.5 mm	3.8 mm	144.78 mm	3.8 mm	2 mm	2 mm	~0.7 mm
Volume	528.22 cm <sup>3</sup>	29.8 cm <sup>3</sup>	1173 cm <sup>3</sup>	29.8 cm <sup>3</sup>	11.5 cm3	11.5 cm <sup>3</sup>	5 cm <sup>3</sup>
Mass	64g	120g	110g	50g	~27g	~27g	10.5g
Beam Width	60°	70°	72°	60°	81°	78°	47°
Band	S-band	S-band	S-band	S-band	S or X-band	S or X-band	S or X-band
Туре	Patch	Patch	Helix	Patch	Patch	Patch	Helix