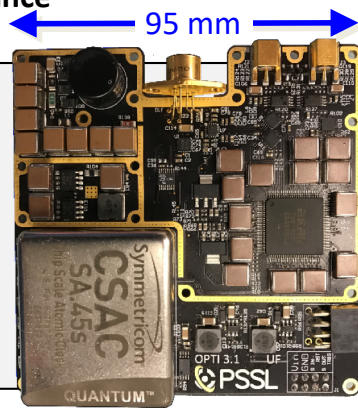


Technology Title: Optical Precision Time-transfer Instrument (OPTI)

Affiliation: University of Florida, Precision Space Systems Lab

Technology Description, Current Performance Metrics, and Performance Goals

- Compact, low power precision time transfer via exchange of nanosecond IR laser pulses
- 100 ps (3 cm) measured time-transfer accuracy
- Chip Scale Atomic Clock: 20 ns drift after 10^4 s
- Ground-to-space or space-to-space links
- Rx: < 5 W peak Tx: < 15 W peak
- Volume: < 2U Mass: < 2 kg
- Ground-to-LEO demo of OPTI: July 2018
- LEO-to-LEO demo with related tech: 2020



Current TRL

TRL 4

TRL By
May 2021

TRL 6

Industry State of the Art Technology Performance

Radio frequency synchronization; high Size, Weight, Power (SWaP)

- GNSS: 3-10 nanoseconds
- TWSTFT: 1-2 nanoseconds

Laser-based synchronization

- Time-transfer by Laser link (T2L2): ~50 ps, 42 W, 5 kg
- ACES-ELT (future): < 25 ps, ISS payload

Technology Development Challenges to Meet TRL Goal

Making the technology rad hard

- Existing technology developed for short (< 1 yr) CubeSat missions in LEO
- For HPD missions, a radiation tolerant version may be needed
- For most components, rad tolerant parts can be found
 - One exception: Time-to-Digital Converter (TDC) that performs the precision time-stamping
 - Solution: The PSSL, with STMD support, has been developing TDCs and related functionalities in software on rad tolerant FPGAs

Precision beam pointing and acquisition still under development, currently TRL 4

Potential HPD Science Application (Optional)

- Synchronization of measurements made by swarms or constellations of microsats or CubeSats
- Example:
 - Particle detectors on a number of small satellites orbiting the Sun
 - Synchronization of the constellation to ~1 ns → relative arrival times of particles at each S/C + relative range between S/C
 - Enables 4D mapping (X, Y, Z, t) of solar activity

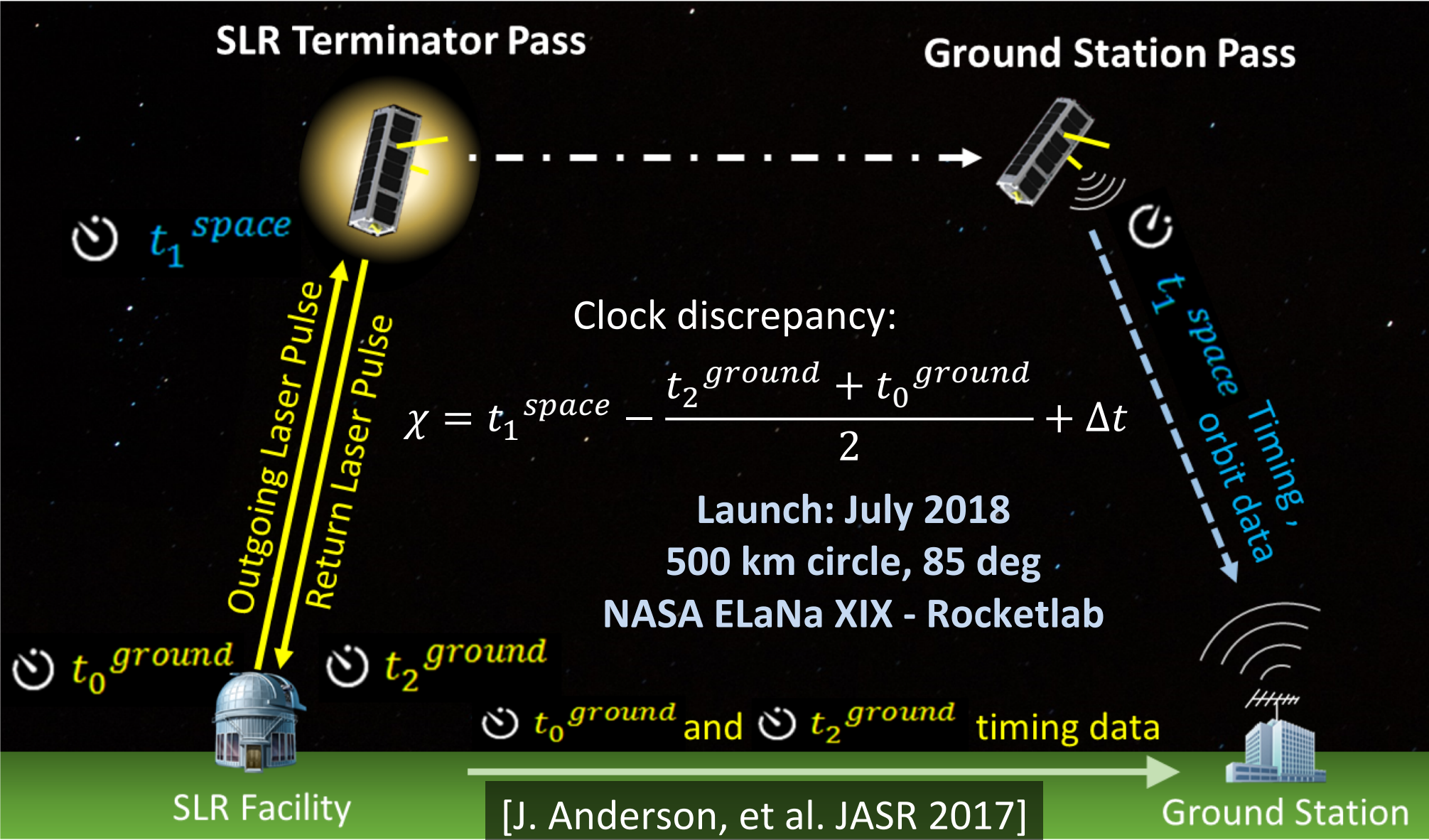
Contact Information

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1-352-392-0614

Additional Comments

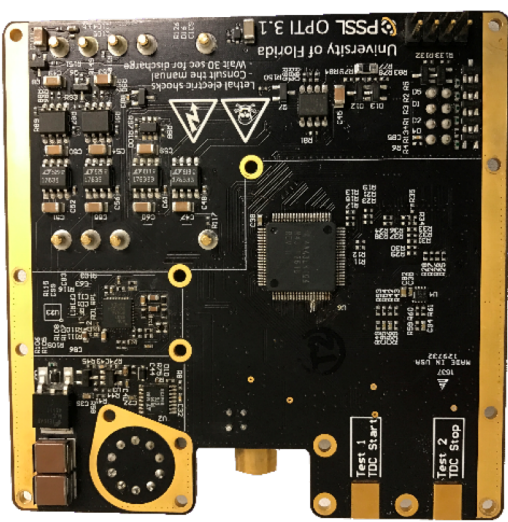
- Additional information on two CubeSat tech demonstration missions (CHOMPTT and CLICK) provided in back-up charts

CHOMPTT CubeSat Mission Concept

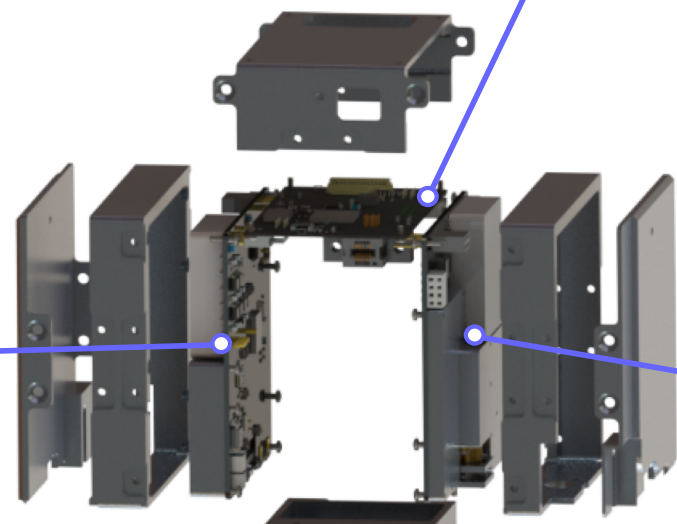


OPTI Flight Payload Assembly

Supervisor
(payload controller)

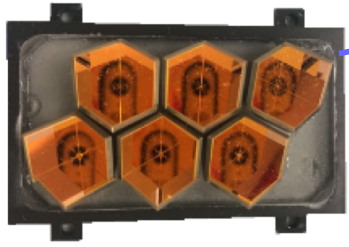


Channel A (back)



Channel B (front)

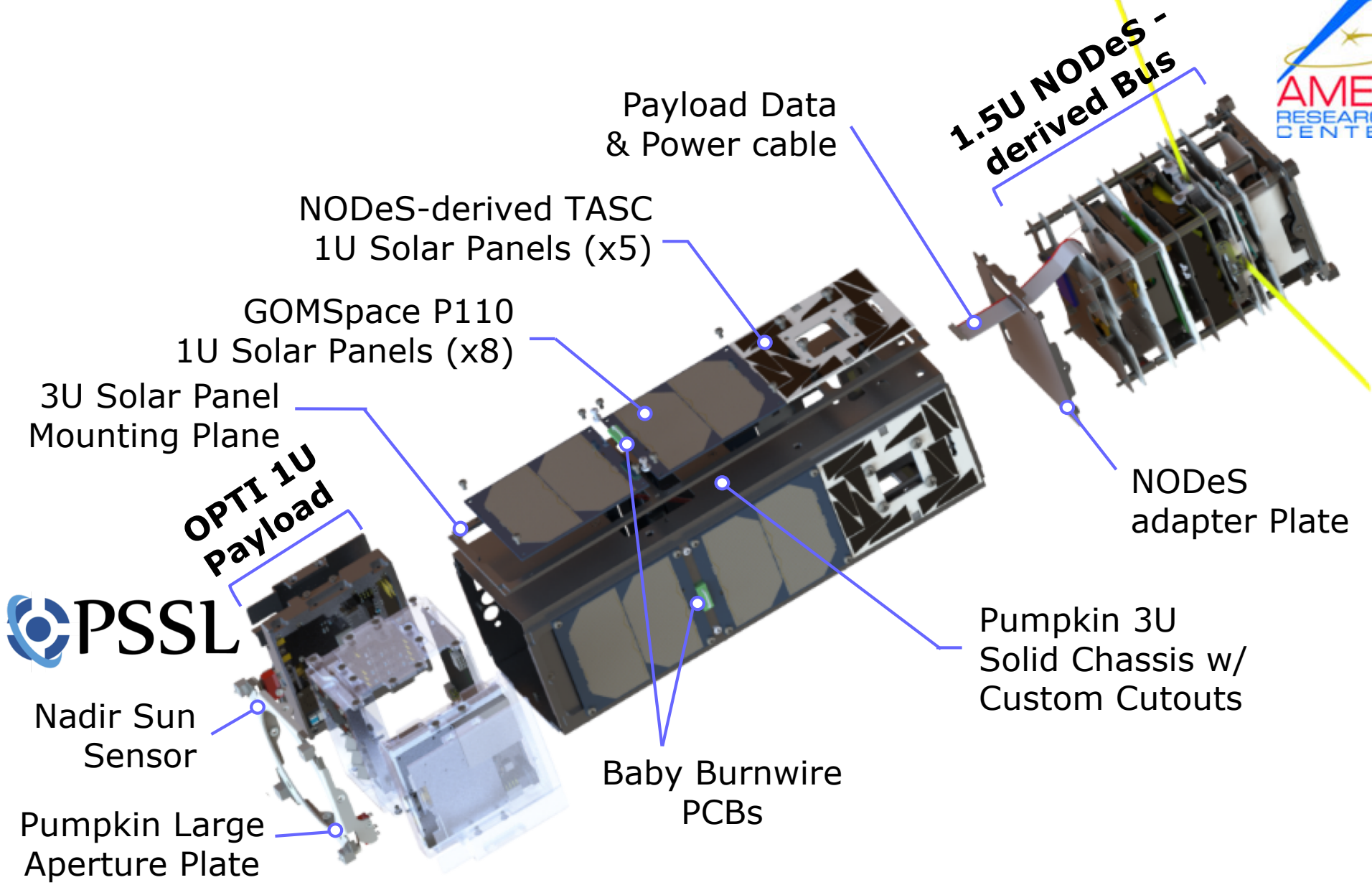
Six, 1 cm retro array



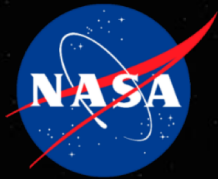
808 nm laser beacons
(4x0.5 W)



CHOMPTT 3U Spacecraft



CubeSat Laser Infrared Crosslink (CLICK)



Two 6U, 15 kg each, P/L < 2 kg
15 W Ave, 30 W peak
Launch in 2020

UF PSSL time-transfer hardware:

- FPGA modulator + CSAC
- TDC-based photoreceiver

NASA ARC-led mission, S/C bus

MIT: Payload PI, Optical front-end,
pointing/tracking, laser comm mod/demod

